

Figure 41. Pressure contours ([7]-Min2).

Figures 40 and 41 exhibit the pressure contours obtained by the [7] scheme, in its Min1 and Min2 variants, respectively. The most severe pressure field is due to the Min2 version of the [7] algorithm.

Figure 42 shows the wall pressure distribution obtained by the [7] in its two variants. They are compared with the experimental results of [42]. As can be observed, the reasonable solution is obtained by the [7] scheme using Min1 limiter. Hence, it is possible to conclude that for the laminar viscous results, the [7] scheme, in its Min1 version, provides the best solution.

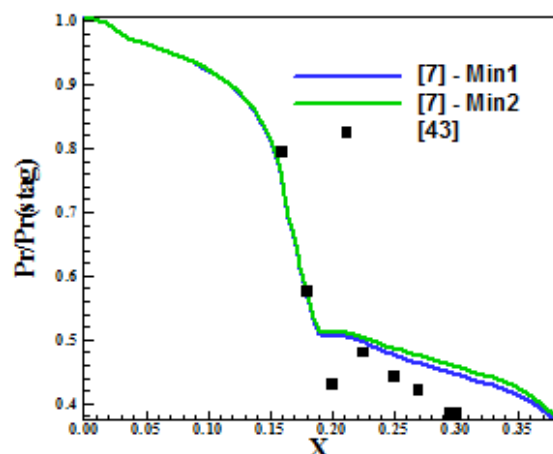


Figure 42. Wall pressure distribution ([7]).

## 11 Conclusion

In the present work, the [7] TVD symmetric, the [9] TVD symmetric, the [12] TVD, and the [13] TVD/ENO schemes are implemented, on a finite volume context and using a structured spatial discretization, to solve the Euler and Navier-Stokes equations in the three-dimensional space. With the exception of [7; 9], all others schemes are high

resolution flux difference splitting ones, based on the concept of Harten's modified flux function. The [7; 9] TVD schemes are symmetric ones, incorporating TVD properties due to the appropriated definition of a limited dissipation function. All schemes are second order accurate in space. An implicit formulation is employed to solve the Euler equations, whereas a time splitting method, an explicit method, is used to solve the Navier-Stokes equations. An approximate factorization in Linearized Nonconservative Implicit LNI form is employed by the [12-13] schemes, whereas an approximate factorization ADI method is employed by the [7; 9] schemes. All algorithms are first order accurate in time. The algorithms are accelerated to the steady state solution using a spatially variable time step, which has demonstrated effective gains in terms of convergence rate ([30-31]). All schemes are applied to the solution of physical problems of the supersonic flows along a ramp and along a compression corner, in the inviscid case, whereas in the laminar viscous case, the supersonic flow along a convergent-divergent nozzle is solved.

The results have demonstrated that the [9] algorithm, with Min2 non-limiter, and [13], in its ENO version, has presented the best solutions in the inviscid ramp and compression corner problems; In the viscous problem, the [7] algorithm, in its Min1 variant, has presented the best solution in the viscous nozzle problem.

This work is the first part of this study, which compares different TVD and ENO algorithms. The next paper will treat more four numerical algorithms based on the Yee's and Yang's works.

## 12 Acknowledgments

The author acknowledges the CNPq by the financial support conceded under the form of a DTI (Industrial Technological Development) scholarship no. 384681/2011-5. He also acknowledges the infrastructure of the ITA that allowed the realization of this work.

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