

Improvement of Sliding Mode Power Control for WECS based on DFIG-Generator

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Abstract: In this work, we are developing a new strategy for controlling wind systems based on the DFIG Generator. The SMC sliding mode technique is based on the principle of Lyapunov stability in order to make a nonlinear system close to linearity. The use of such a technique with an improvement in regulators to eliminate the Chattering phenomenon shows a great improvement in the performance of wind systems. which is based on performance estimators to improve the quality of the system. The energy quality at the output of the wind system will be injected into the distribution network according to international standards. The proposed model is validated on the Matlab & Simulink environment to test trajectory tracking and robustness.

Key-Words: Wind power generator system; SMC Control; PMSG; MPPT; MATLAB/SIMULINK)

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1. Introduction

In the recent decades, the scarcity of fossil fuels and their impact on the environment have led to increase the usage of alternative energy resources, such as wind and solar energy. Hence, the growing interest about the electrical energy produced from the wind power from the industries.

Today, most installed wind turbines are equipped with asynchronous double-feed machines (DFIG). Indeed, this type of machine has several advantages such as their ability to produce electrical energy at variable speed and extract the maximum possible power, Where The interest of variable speed for a wind turbine is to be able to operate on a large range of wind speeds, and to be able to derive the maximum power possible for each wind speed. Thus their accessibility to the stator and the rotor which offers the opportunity to have several degrees of freedom to properly control the transfer of power and the power factor. Where we can have the rotor circuit controlled by a low power converter relatively to the stator. DFIG is subject to many constraints, such as the effects of parametric uncertainties (due to overheating, saturation, etc.) and the disturbance of the speed variation, which could divert the system from its optimal functioning. purpose of solving this kind of problems we have referred to the use of the command by sliding mode.

2. Modeling of Wind System Based on DFIG

The conversion chain includes in series a speed multiplier to increase the speed of rotation to about 1500 rpm, a doubly fed induction generator(DFIG) operating at variable speed, Three-phase converters adjust the frequency of the wind turbine to that of the electricity grid to which it is connected (50 Hz in Morocco).

The transformation of the power of the aerogenerator into kinetic energy then into mechanical energy of rotation is done in two parts: at the turbine rotor (primary shaft), which captures part of the kinetic energy of the wind present to

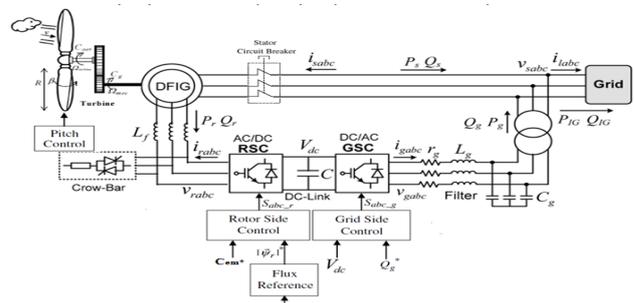


Figure 1. Architecture of the Control

1.1. Wind-Turbine model

The model of the turbine is modeled from the following system of equations: [7-8]:

$$P_{incident} = \frac{1}{2} \cdot \rho \cdot S \cdot v^3 \quad (1)$$

$$P_{extracted} = \frac{1}{2} \cdot \rho \cdot S \cdot C_p(\lambda, \beta) \cdot v^3 \quad (2)$$

$$\lambda = \frac{\Omega_t \cdot R}{v} \quad (3)$$

$$C_p^{max}(\lambda, \beta) = \frac{16}{27} \approx 0.593 \quad (4)$$

$$C_p(\lambda, \beta) = c_1 \cdot \left(c_2 \cdot \frac{1}{A} - c_3 \cdot \beta - c_4 \right) \cdot e^{-c_5 \frac{1}{A}} + c_6 \cdot \lambda \quad (5)$$

$$\frac{1}{A} = \frac{1}{\lambda + 0.08 \cdot \beta} - \frac{0.035}{1 + \beta^3} \quad (6)$$

$$C_{al} = \frac{P_{eol}}{\Omega_t} = \frac{1}{2} \cdot \rho \cdot S \cdot C_p(\lambda, \beta) \cdot v^3 \cdot \frac{1}{\Omega_t} \quad (7)$$

$$J = \frac{J_{tur}}{G^2} + J_g \quad (8)$$

$$J \frac{d\Omega_{mec}}{dt} = C_{mec} = C_{ar} - C_{em} - f \cdot \Omega_{mec} \quad (9)$$

S : the area swept by the pales of the turbine [m^2]
 ρ : the density of the air ($\rho = 1.225 kg / m^3$ at atmospheric pressure). v : wind speed [m/s]. $C_p(\lambda, \beta)$: the power coefficient. λ : the specific speed. β : the angle of orientation of the blades. $P_{extracted}$: the power extracted of the turbine.

Ω_t : Rotational speed of the turbine. C_{al} : Torque on the slow axis (turbine side). J_{tur} : turbine inertia. J_g : inertia of the generator. Ω_{mec} : Mechanical speed of DFIG. C_{ar} : Aerodynamic torque on the fast axis of the turbine

The Fig.2 shows the evolution of the power coefficient as a function of λ for different values of β . A coefficient of maximum power of $C_p=0.564$ is obtained for a speed ratio λ which is (3) (λ_{opt}). Fixing β and λ respectively to their optimal values, the wind system provides optimal power.

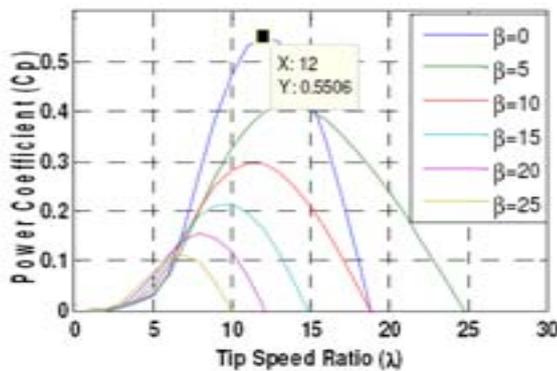


Figure 2. Curves of the power coefficient

1.2. Maximization of power control without speed control.

while the variation of the wind speed in steady state is low compared to the electrical time constants of the system, we assume that the speed of rotation of the DFIG is fixed and neglecting the effect of the viscous torque f , the dynamic equation of the turbine becomes (12); from (12) we obtain the static equation describing the stationary state of the turbine (13); The reference electromagnetic torque is determined from an estimate of the aerodynamic torque given by (14), we obtain equation (15); The orientation angle of the blades β is assumed to be constant and the estimated speed of the turbine is calculated from the mechanical speed (16); The estimated wind speed is given by equation (17); on the Base of the previous equations, we can then write the equation of the reference electromagnetic couple (18).

$$J_{tot} \cdot \frac{d\Omega_{mec}}{dt} = C_g - C_{em} - f \cdot \Omega_{mec} = 0 \quad (10)$$

$$C_g = C_{em} = \frac{C_{aero}}{G} \quad (11)$$

$$C_{aero_{est}} = \frac{1}{2 \cdot \Omega_{t_{est}}} \cdot C_{p_{max}}(\lambda_{opt}, \beta) \cdot \rho \cdot \pi \cdot R^2 \cdot v_{est}^3 \quad (12)$$

$$C_{em_{ref}} = \frac{C_{aero_{est}}}{G} \quad (13)$$

$$\Omega_{t_{est}} = \frac{\Omega_{mec}}{G} \quad (14)$$

$$v_{est} = R \cdot \frac{\Omega_{est}}{\lambda_{opt}} \quad (15)$$

$$C_{em_{ref}} = \frac{\rho \cdot \pi \cdot R^5 \cdot C_{p_{max}}(\lambda_{opt}) \cdot \Omega_{mec}^2}{2 \cdot \lambda_{opt}^3 \cdot G^3} \quad (16)$$

1.3. DFIG Model

The equations of the stator voltages V_s (d, q) and the rotor V_r (d, q) of the DFIM in the reference of Park are written as follows: [9, 13]:

Voltages at the stator:

$$\begin{cases} V_{sd} = R_s \cdot i_{sd} + \frac{d\psi_{sd}}{dt} - \omega_s \cdot \psi_{sq} \\ V_{sq} = R_s \cdot i_{sq} + \frac{d\psi_{sq}}{dt} + \omega_s \cdot \psi_{sd} \end{cases} \quad (17)$$

• Voltages at the stator:

$$\begin{cases} V_{rd} = R_r \cdot i_{rd} + \frac{d\psi_{rd}}{dt} - \omega_r \cdot \psi_{rq} \\ V_{rq} = R_r \cdot i_{rq} + \frac{d\psi_{rq}}{dt} + \omega_r \cdot \psi_{rd} \end{cases} \quad (18)$$

With: $\omega_s - \omega_r = p \cdot \omega$

The magnetic equations are expressed by the flux expressions in the reference (d, q).

• Flux at the stator :

$$\begin{cases} \psi_{sd} = L_s \cdot \dot{\psi}_{sd} + M \cdot i_{rd} \\ \psi_{sq} = L_s \cdot \dot{\psi}_{sq} + M \cdot i_{rq} \end{cases} \quad (19)$$

• Flux at the rotor :

$$\begin{cases} \psi_{rd} = L_r \cdot \dot{\psi}_{rd} + M \cdot i_{sd} \\ \psi_{rq} = L_r \cdot \dot{\psi}_{rq} + M \cdot i_{sq} \end{cases} \quad (20)$$

With: $M=M_{sr}=M_{rs}$

The electromagnetic torque is expressed as a function of the currents and the flows by:

$$\begin{cases} C_{em} = p \cdot (\psi_{sd} \cdot i_{sq} + \psi_{sq} \cdot i_{sd}) \\ C_{em} = p \cdot (\psi_{rq} \cdot i_{rd} - \psi_{rd} \cdot i_{rq}) \end{cases}$$

The fundamental equation of dynamics is:

$$C_{em} = C_r + J \cdot \frac{d\Omega}{dt} + f \cdot \Omega \quad (21)$$

With:

$V_{s(d,q)}, V_{r(d,q)}$: Stator and rotor voltages in the reference of Park.

$I_{s(d,q)}, I_{r(d,q)}$: Stator and rotor currents in the reference of Park.

$\psi_{s(d,q)}, \psi_{r(d,q)}$: Stator and rotor flux in the reference of Park.

R_s, R_r, L_s, L_r : Stator and rotor resistances and inductances.

M : Mutual inductance.

C_r, C_{em} : Load torque and Electromagnetic torque.

Ω : Rotation speed of the machine ($\omega=p \cdot \Omega$).

J, f : Moment of inertia and Coefficient of viscous friction.

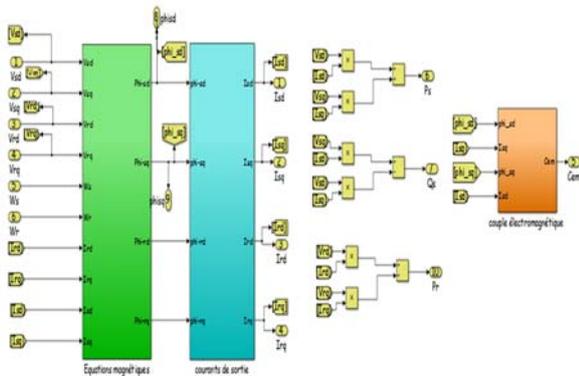


Figure 4. Simulink Model of the DFIG

1.4. Model of the two-level voltage inverter:

The two-level voltage inverter controls the powers (active and reactive) exchanged between the generator and the network. In order to facilitate the modeling of the power converter, Each inverter arm consists of two semiconductor switches assumed to be perfect and functioning in a complementary manner. It can be used in inverter mode or in rectifier mode (the modeling remains valid for both cases) .The output voltage of the two three-phase

converters GSC (inverter) and RSC (rectifier) is controlled by sinusoidale pulse width modulation technique (PWM) which allows simultaneous adjustment of the frequency and the RMS value of the output voltage.

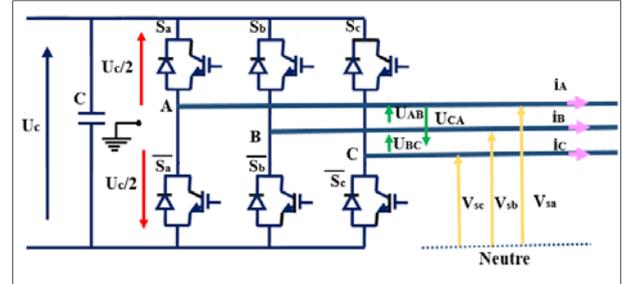


Figure 5. Diagram of the two-level inverter

The Sa, Sb, Sc is the state of the upper switches of each arm of the inverter.

- The expression of the simple voltages is presented by the following system:

$$\begin{cases} v_A = \frac{1}{3}(U_{AB} - U_{CA}) = \frac{1}{3}(2 \cdot v_A - v_B - v_C) \\ v_B = \frac{1}{3}(U_{BC} - U_{AB}) = \frac{1}{3}(2 \cdot v_B - v_A - v_C) \\ v_C = \frac{1}{3}(U_{CA} - U_{BC}) = \frac{1}{3}(2 \cdot v_C - v_A - v_B) \end{cases} \quad (22)$$

- The matrix form of simple tensions becomes:

$$\begin{bmatrix} v_{AO} \\ v_{BO} \\ v_{CO} \end{bmatrix} = \frac{1}{3} \begin{bmatrix} 2 & -1 & -1 \\ -1 & 2 & -1 \\ -1 & -1 & 2 \end{bmatrix} \cdot \begin{bmatrix} v_{AO} \\ v_{BO} \\ v_{CO} \end{bmatrix} \quad (23)$$

We associated with Each arm of the inverter a binary command value Si, where i = a, b, c:

$$\begin{bmatrix} v_{AO} \\ v_{BO} \\ v_{CO} \end{bmatrix} = \frac{U_{CC}}{2} \begin{bmatrix} S_a \\ S_b \\ S_c \end{bmatrix} \quad (24)$$

We replace (25) in (24), we get:

$$\begin{bmatrix} v_A \\ v_B \\ v_C \end{bmatrix} = \frac{U_{CC}}{6} \cdot \begin{bmatrix} 2 & -1 & -1 \\ -1 & 2 & -1 \\ -1 & -1 & 2 \end{bmatrix} \cdot \begin{bmatrix} S_a \\ S_b \\ S_c \end{bmatrix} \quad (25)$$

The single voltages of the inverter become proportional to the states of the control quantities of the switches (Sa, Sb, Sc).

1.5. DC-Bus model

The DC bus interconnects the two converters of the wind system (RSC and GSC). The latter allows the transfer of power between two sources at different frequencies. It is modeled by equations (27);

$$\begin{cases} W_{dc} = \int P_c \cdot dt = \frac{1}{2} \cdot C \cdot V_{dc}^2 \\ \frac{dV_{dc}^2}{dt} = \frac{2}{C} (P_f - P_r) \end{cases} \quad (26)$$

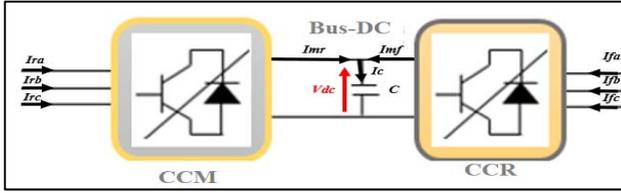


Figure 6. Diagram of the continuous bus

1.6. RL Filter Model

The intermediate filter used is of the RL type. The currents passed between the GSC converter and the network are imposed by the coils constituting the low pass filter. The expressions of the voltages at the terminals of the filters in the the park referential are presented by (28).

$$\begin{cases} v_{df} = -R_f \cdot I_{df} - L_f \cdot \frac{dI_{df}}{dt} + \omega_s \cdot L_f \cdot I_{qf} \\ v_{qf} = -R_f \cdot I_{qf} - L_f \cdot \frac{dI_{qf}}{dt} - \omega_s \cdot L_f \cdot I_{df} + v_s \end{cases} \quad (27)$$

3. Sliding mode controller

The basic idea of sliding mode control design is first to attract the states of the system to a suitably selected region, and then to design a control law that will always keep the system in that region. Where the design of the sliding mode control algorithm is defined by three complementary steps:

3.1. Choice of sliding surface

For a non-linear system presented in the following form (29), where $A(x,t)$ and $B(x,t)$ are two continuous and uncertain non linear functions assumed to be bounded.

$$\begin{cases} \dot{x}(t) = A(x,t) + B(x,t)u(t) ; x \in R^n, u \in R & (28) \\ (x) = \left(\frac{d}{dt} + \delta\right)^{n-1} * e(x) & (29) \\ e(x) = x^d - x & (30) \\ x = [x, \dot{x}, \dots, x^{n-1}]^T ; x^d = [x^d, \dot{x}^d, \dots]^T \end{cases}$$

3.2. Convergence and existence conditions

To make the surface attractive and invariant, we returned to the second theorem of LYAPUNOV where the scalar function is defined positive by (32). The derivative of this function gives (33), and to guarantee the existence of the sliding mode, where the sliding variable $S(x, t)$ tends towards zero, it suffices to ensure that (33) is defined negative (34).

$$V(x) = \frac{1}{2} \cdot S(x)^2 \quad (31)$$

$$\dot{V}(x) = S(x)S'(x) \quad (32)$$

$$S(x)S'(x) < 0 \quad (33)$$

3.3. Determination of the Law of Control

The control law is defined by the relation (35), With: u^+ and u^- are continuous functions (36) where

$u^- \neq u^+$. The control by sliding modes is composed of two terms: u_{eq} : the equivalent control vector, u_n : The stabilizing command.

$$u = u_{eq} + u_n \quad (34)$$

$$u = \begin{cases} u^+(x) & \text{si } S(x, t) > 0 \\ u^-(x) & \text{si } S(x, t) < 0 \end{cases} \quad (35)$$

✓ Determination of the equivalent command

u_{eq} :

$$S(\dot{x}) = \frac{dS}{dt} = \frac{\partial S}{\partial x} \cdot \frac{\partial x}{\partial t} \quad (36)$$

$$S(\dot{x}) = \frac{dS}{dt} = \frac{\partial S}{\partial x} (A(x, t) + B(x, t)(u_{eq} + u_n)) \quad (37)$$

$$S(\dot{x}) = \frac{dS}{dt} = \frac{\partial S}{\partial x} (A(x, t) + B(x, t)u_{eq}) + \frac{\partial S}{\partial x} B(x, t)u_n \quad (38)$$

$$u_{eq}(t) = \left(-\frac{\partial S}{\partial x} A(x, t)\right) \left(\frac{\partial S}{\partial x} B(x, t)\right)^{-1} \quad (39)$$

✓ Determination of the basic discontinuous command u_n :

The simplest discontinuous command u_n is given by (45), where K is the command gain. This type of control has a drawback known by "CHATTERING". To solve this problem in this case we replaced the "SIGN" function by the "SAT" function:

$$u_n = K \cdot \text{sign}(S(x)) \quad (40)$$

$$\text{Sat}(S) = \begin{cases} \text{Sat}(S) = 1 & \text{si } S > \varepsilon \\ \text{Sat}(S) = -1 & \text{si } S < -\varepsilon \\ \text{Sat}(S) = \frac{S}{\varepsilon} & \text{si } |S| < \varepsilon \end{cases} \quad (41)$$

4. Application Of The Sliding Mode Command To The DFIG

4.1. Control of the converter on the DFIG (RSC) side and on the Grid side (GSC)

Considering the sliding surface proposed by SLOITINE (30), For $n = 1$; the sliding surface of the active and reactive power is given by (47), where P_{sref} and Q_{sref} are the references of stator powers (active and reactive) of DFIG and Q_{fref} and P_{fref} are the references of powers (reactive and active) of RL filter.

$$\begin{cases} S(P_s) = e_1 = P_{sref} - P_s \\ S(Q_s) = e_2 = Q_{sref} - Q_s \\ S(P_f) = e_3 = P_{fref} - P_f \\ S(Q_f) = e_4 = Q_{fref} - Q_f \end{cases} \quad (42)$$

$$\begin{cases} S(\dot{P}_s) = \dot{e}_1 = \dot{P}_{sref} - \dot{P}_s \\ S(\dot{Q}_s) = \dot{e}_2 = \dot{Q}_{sref} - \dot{Q}_s \\ S(\dot{P}_f) = \dot{e}_3 = \dot{P}_{fref} - \dot{P}_f \\ S(\dot{Q}_f) = \dot{e}_4 = \dot{Q}_{fref} - \dot{Q}_f \end{cases} \quad (43)$$

With:

$$\begin{cases} \dot{P}_s = -\frac{v_s \cdot M}{L_s} \dot{I}_{rq} \\ \dot{Q}_s = \frac{v_s^2}{\omega_s \cdot L_s} - \frac{v_s \cdot M}{L_s} \dot{I}_{rd} \\ \dot{I}_{rd} = \frac{V_{rd}}{L_r \cdot \sigma} - \frac{R_r}{L_r \cdot \sigma} \cdot I_{rd} + \omega_r \cdot I_{rq} \\ \dot{I}_{rq} = \frac{V_{rq}}{L_r \cdot \sigma} - \frac{R_r}{L_r \cdot \sigma} \cdot I_{rq} - \omega_r \cdot I_{rd} - \omega_r \cdot \frac{M \cdot V_s}{L_r \cdot L_s \cdot \sigma \cdot \omega_s} \end{cases} \quad (44)$$

We replace each term by its expression given by (49), the derivative of the sliding surface becomes as follows (50) (51):

$$\begin{cases} \dot{e}_3 = \dot{P}_{sref} + \frac{V_s \cdot R_f}{L_f} I_{qf} + \frac{V_s (V_{fqueq} + V_{fqn})}{L_f} + \omega_s \cdot V_s \cdot I_{df} - \frac{V_s^2}{L_f} \\ \dot{e}_4 = \dot{Q}_{sref} - \frac{V_s \cdot R_f}{L_f} I_{df} - \frac{V_s (V_{fdeq} + V_{fdn})}{L_f} + \omega_s \cdot V_s \cdot I_{qf} \end{cases} \quad (46)$$

According to the sliding mode and the permanent mode, we have (54) (56); By replacing the equation ((54) in (52)) and ((56) in (53)), the expression of the equivalent V_{eq} command becomes (57) (58); the stabilizing command is given by equation (59) (60); finally the expression of the total order (V_{rd} , V_{rq}) and (V_{fd} , V_{fq}) becomes (61) (62).

$$\begin{cases} e_{1,2} = 0 \\ \dot{e}_{1,2} = 0 \\ V_{rdn} = V_{rqn} = 0 \end{cases} \quad (47)$$

$$\begin{cases} e_{3,4} = 0 \\ \dot{e}_{3,4} = 0 \\ V_{fdn} = V_{fqn} = 0 \end{cases} \quad (48)$$

$$\begin{cases} V_{rdeq} = -\frac{L_r \cdot L_s \cdot \sigma}{M \cdot V_s} \dot{Q}_{sref} + R_r \cdot I_{rd} - \omega_r \cdot L_r \cdot \sigma \cdot I_{rq} \\ V_{rqueq} = -\frac{L_r \cdot L_s \cdot \sigma}{M \cdot V_s} \dot{P}_{sref} + R_r \cdot I_{rq} + \omega_r \cdot L_r \cdot \sigma \cdot I_{rd} + \omega_r \cdot V_s \cdot \frac{M}{L_s \cdot \omega_s} \\ V_{fdeq} = \frac{L_f}{V_s} \dot{Q}_{sref} - R_f \cdot I_{df} + L_f \cdot \omega_s \cdot I_{qf} \\ V_{fqeq} = \frac{L_f}{V_s} \dot{P}_{sref} - R_f \cdot I_{qf} - L_f \cdot \omega_s \cdot I_{df} + V_s \end{cases} \quad (50)$$

$$\begin{cases} V_{rdn} = K_{rdn} \cdot Sat(e_1) \\ V_{rqn} = K_{rqn} \cdot Sat(e_2) \\ V_{fdn} = K_{fdn} \cdot Sat(e_4) \\ V_{fqn} = K_{fqn} \cdot Sat(e_3) \end{cases} \quad (51)$$

$$\begin{cases} V_{rd} = -\frac{L_r \cdot L_s \cdot \sigma}{M \cdot V_s} \cdot \dot{Q}_{sref} + R_r \cdot I_{rd} - \omega_r \cdot L_r \cdot \sigma \cdot I_{rq} + K_d sat(e_2) \\ V_{rq} = -\frac{L_r \cdot L_s \cdot \sigma}{M \cdot V_s} \cdot \dot{P}_{sref} + R_r \cdot I_{rq} + \omega_r \cdot L_r \cdot \sigma \cdot I_{rd} + \omega_r \cdot M \cdot \frac{V_s}{L_s \cdot \omega_s} + K_d sat(e_1) \end{cases} \quad (52)$$

$$\begin{cases} V_{fd} = -\frac{L_f}{V_s} \cdot \dot{Q}_{fref} - R_f \cdot I_{df} + \omega_s \cdot L_f \cdot I_{qf} + K_d sat(e_4) \\ V_{fq} = -\frac{L_f}{V_s} \cdot \dot{P}_{fref} - R_f \cdot I_{qf} - \omega_s \cdot L_f \cdot I_{df} + V_s + K_d sat(e_3) \end{cases}$$

5. Simulation & Results

To verify the performance and stability of the control system by SMC control, the DFIG is subject to two robustness tests (the Tracking and Regulation Tests for SMC and the robustness tests regarding the variation parameters)

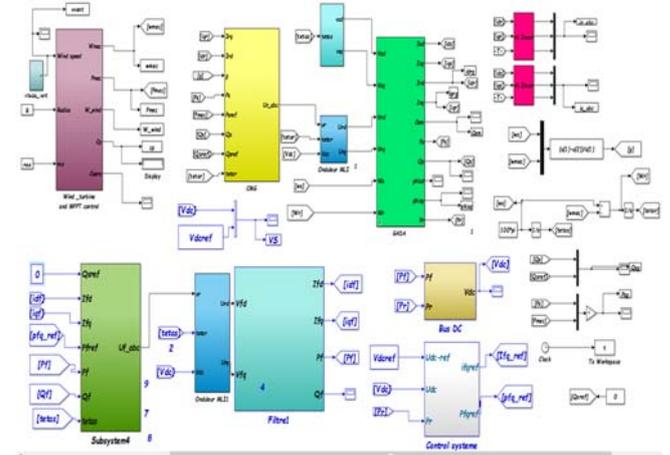
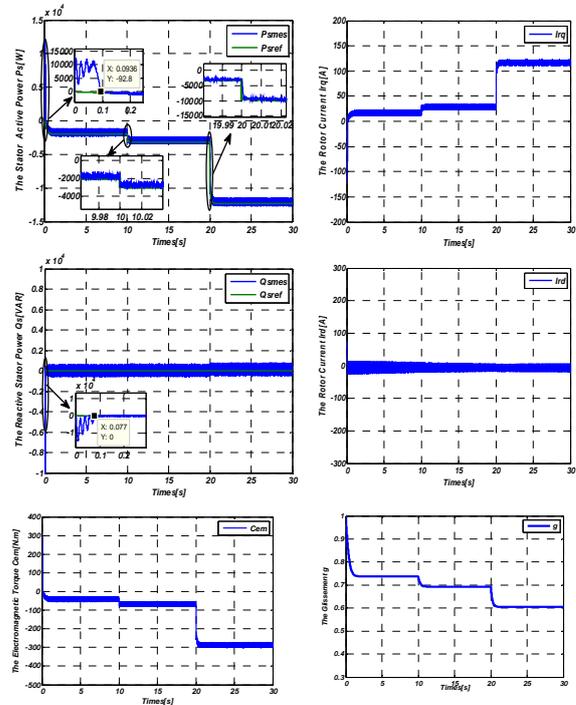


Figure 6. Simulation scheme of the sliding mode control on MATLAB and Simulink environment

5.1. Tracking and Regulation for SMC

In this test, we considered the aerodynamic power according to the MPPT as a reference of active power, and zero as reference for reactive power.

✓ Test with constant speed (runc speed):



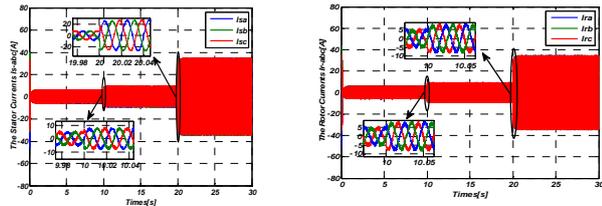


Figure 7. Results of the control by sliding mode of the DFIG

According to the curves illustrated in figure (8) we note that the power references are perfectly followed by the DFIG generator, where the continuous bus DC perfectly follows its reference value without overshoot and with a low response time, and the rotor currents I_r -abc and the three-phase stator currents I_s -abc generated by the GFID, are of the sinusoidal form with a fixed frequency 50Hz for the stator currents. This implies clean energy without harmonics exchanged between DFIG and the electricity network.

✓ **Test with variable speed:**

During this test the wind profile illustrated in figure (8) was applied to the DFIG.

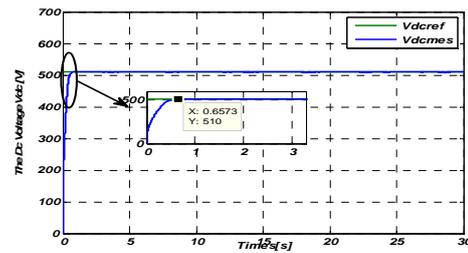
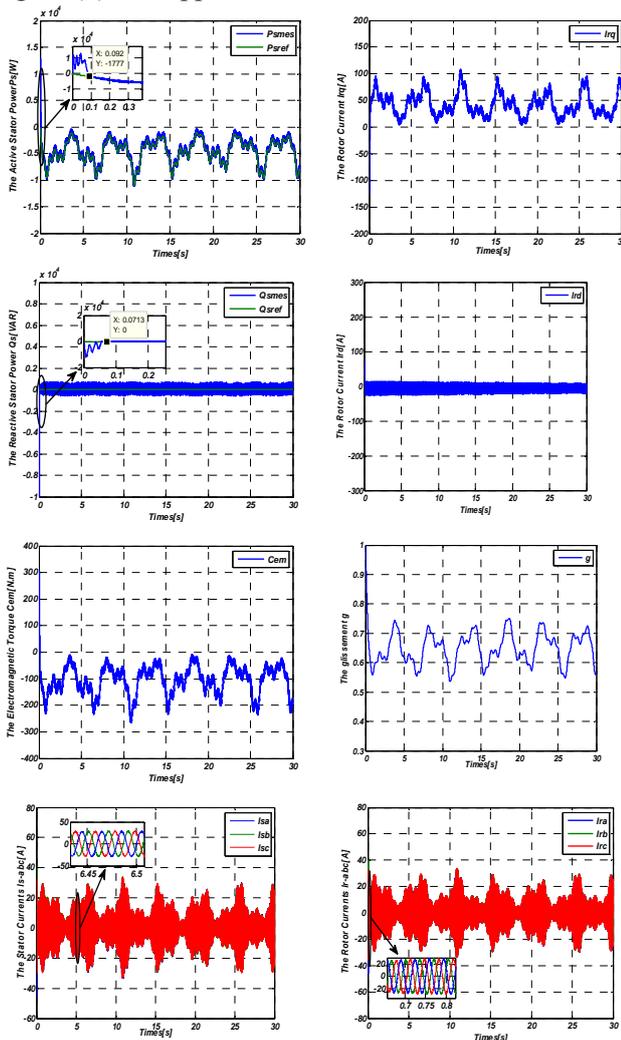


Figure 8. Results of the control by sliding mode at variable

According to the curves illustrated in the figure (10) we notice a good behavior of the machine in spite of the variation of the wind, where the generator follows the references of the powers without overshoot and with an almost zero error. And the electromagnetic torque of the DFIG varies according to the wind speed, and proportional to the active stator power generated. We can notice that in spite of the variations of the wind, the stator current I_s -abc remains sinusoidal with a fixed frequency 50Hz equivalent to that of the network. The DC bus voltage shows that it follows its reference value quickly without overshooting with a small static error. the slip value g is negative this implies that the functioning of the DFIG is in hypo-synchronous

✓ **Robustness Tests**

In this test, we varied the parametric variables of the DFIG (resistance R_r and R_s) by + 50% of their nominal values.

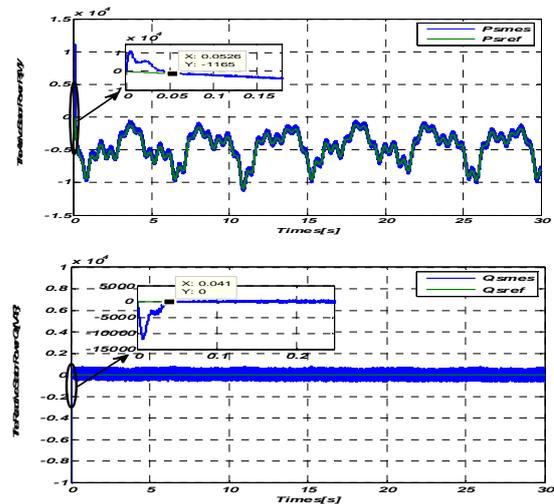


Figure 9. Robustness test with variation of resistances R_r and R_s for a variable wind speed
 It is noted that the decoupling between these two powers remains achievable with the same response time despite the variation of the resistance's R_r and R_s . This shows the robustness of the control by sliding mode facing this variation.

6. Conclusion

The goal of this work is devoted to improving the power generated by a wind system by using different control algorithms to ensure a control structure achieving the best value simplicity, performance and stability. we started with MPPT control to ensure maximum power extraction. then, we implemented the sliding mode control which presents an optimal solution to the stability of nonlinear systems. Afterwards, we carried out the simulation in the Matlab / Simulink environment. and finally we applied a wind profile to the DFIG to analyze the behavior of the latter in dynamic regime. The results show that the sliding mode control system can be considered as an optimal solution in the field of renewable energies.

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